

Investigating the potential of turbulence measurements with the RPAS SUMO



Line Båserud

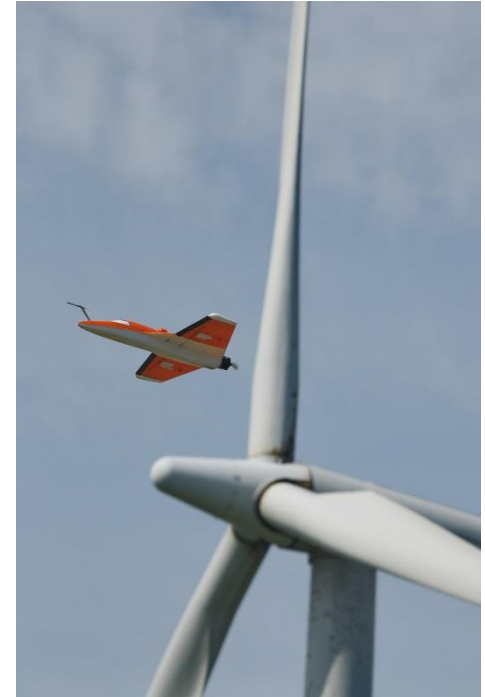


UNIVERSITY OF BERGEN
GEOPHYSICAL INSTITUTE

Outline

- Motivation
- The SUMO system
 - The 5-hole probe
- Wind tunnel testing
 - Comparison to hot-wire anemometer
 - Tubing length variation
- Calculation methods
 - 3D turbulent wind vector
- BLLAST
 - Comparison to 60 m tower (TKE and u^*)
 - TKE profiles from SUMO
- Summary
- Future work

- New sensors available for RPAS
- RPAS: small, inexpensive and flexible
- Operation close to the ground
- Operation in and around wind farms
- SUMO (GFI)



- Wind tunnel experiments (2013)
- Two SUMO turbulence datasets available:

BLLAST field campaign
(2011)



<http://Map-France.com>

Nøjsomheds Odde
wind farm (2011)



Giebel et.al (2011)

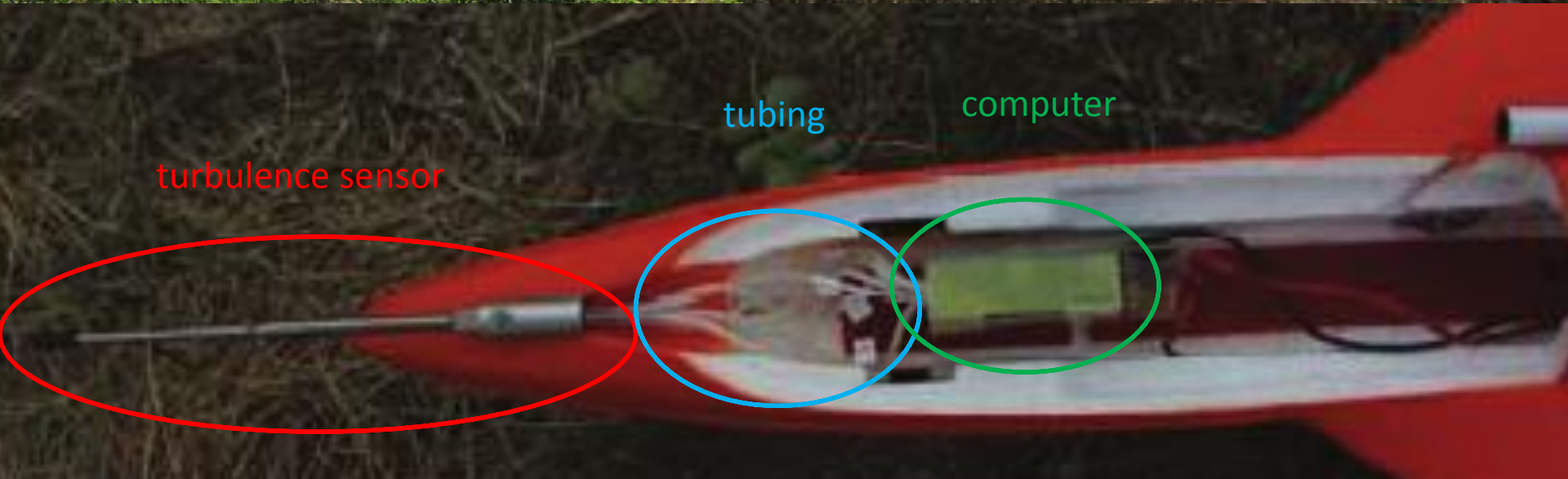
The 5-hole probe (OTF)

- Newly integrated sensor
- 3D turbulent wind
- 100 Hz temporal resolution
- Static and differential pressure

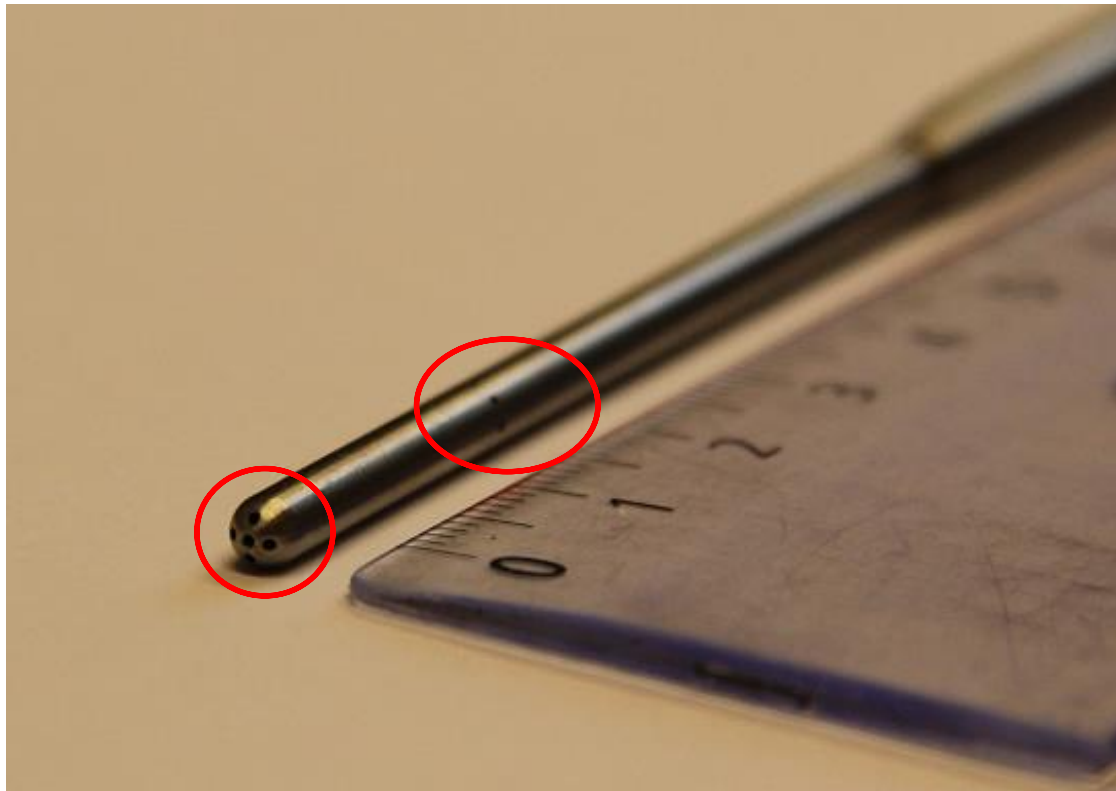


Reuder et. al (2012)



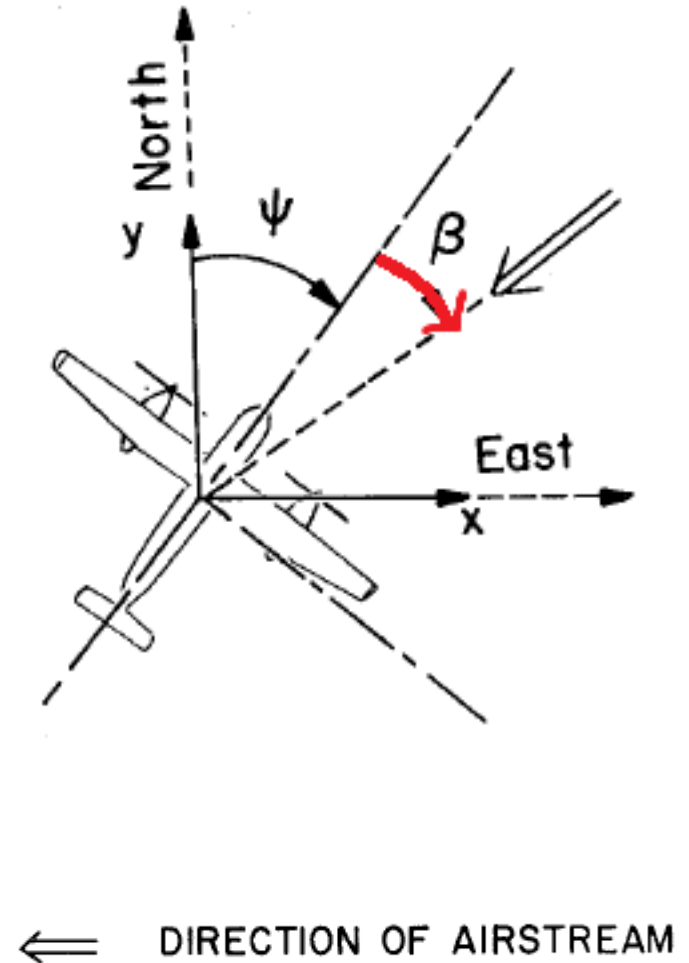
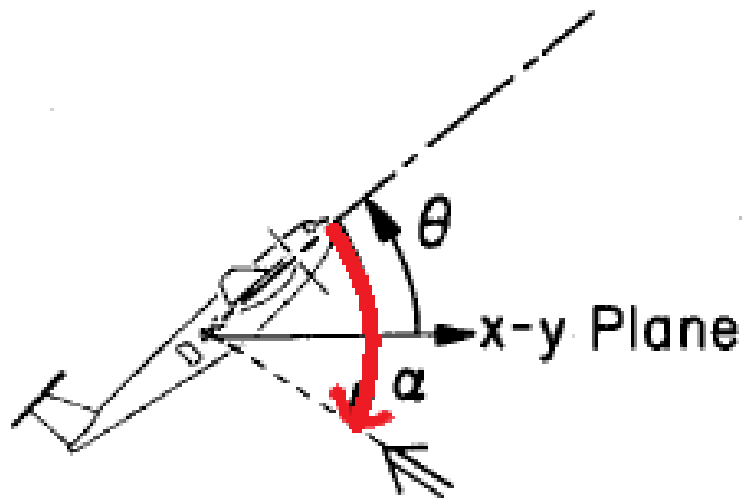


Figures from Reuder et. al (2012) and Reuder & Jonassen (2012)



Output from the 5-hole probe

- True airspeed, U_a
- Angle of attack, α
- Angle of sideslip, β
- Altitude



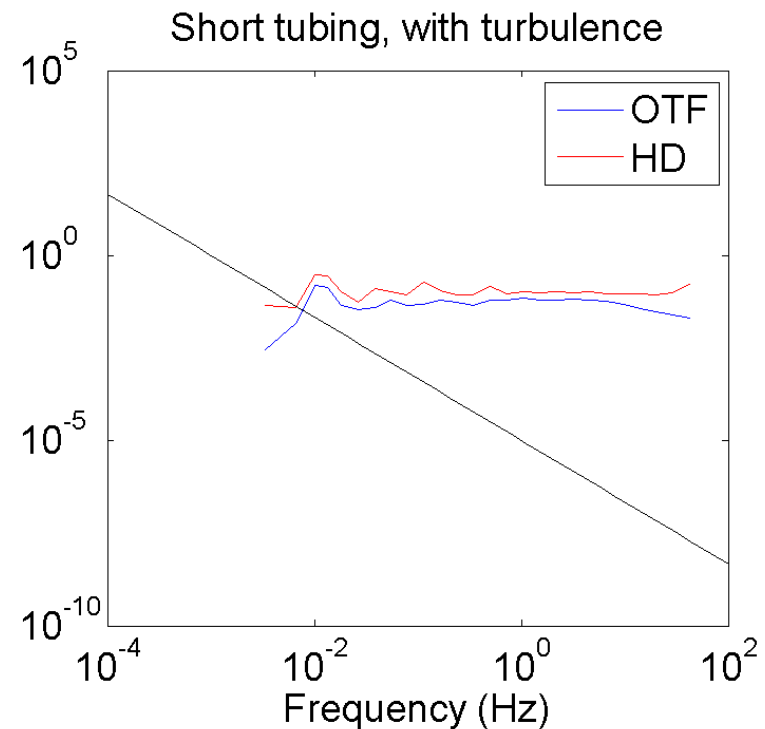
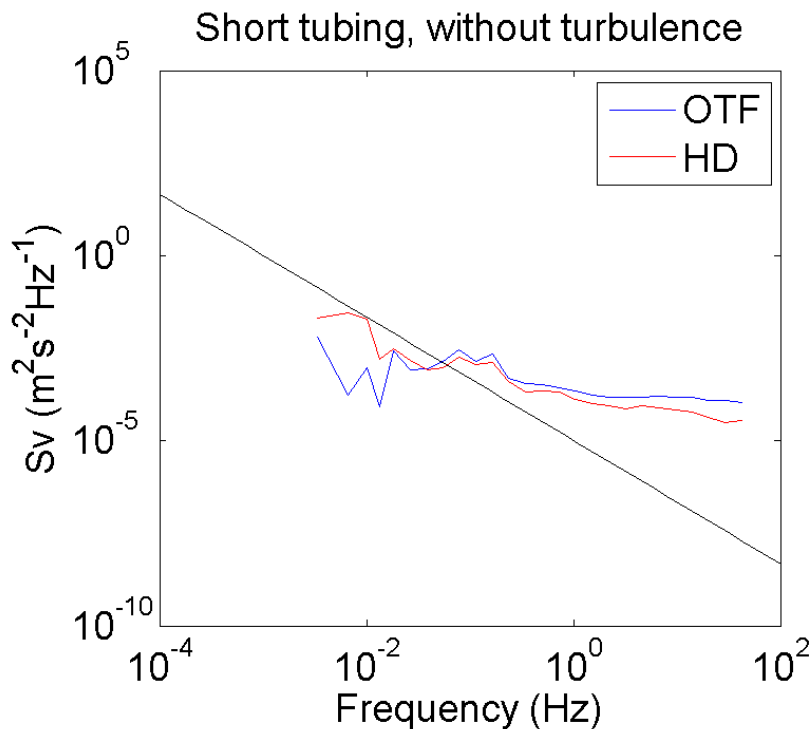
Wind tunnel experiments

- Comparison to hot-wire anemometer (HD)
- Tubing length variation



Comparison to hot-wire anemometer

- OTF responds to energy shift
- Similar response as the HD
- Difference in variability



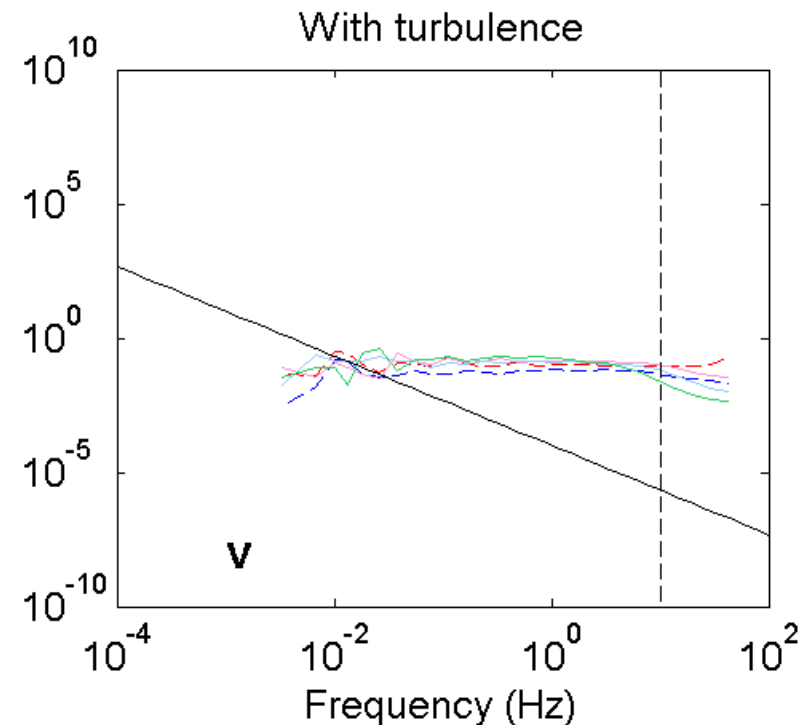
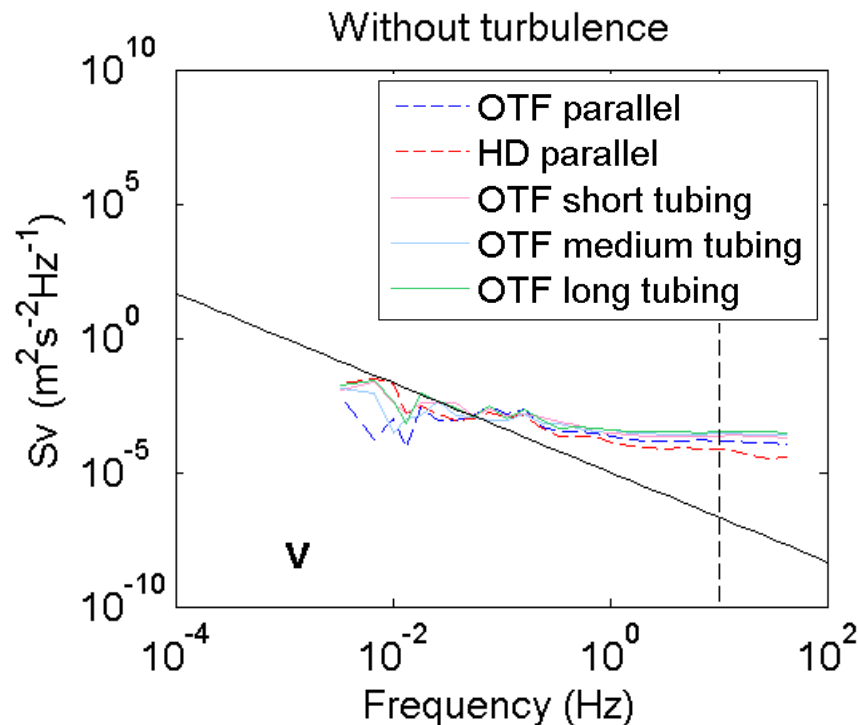
Tubing length variation

- 15 cm
- 30 cm
- 90 cm



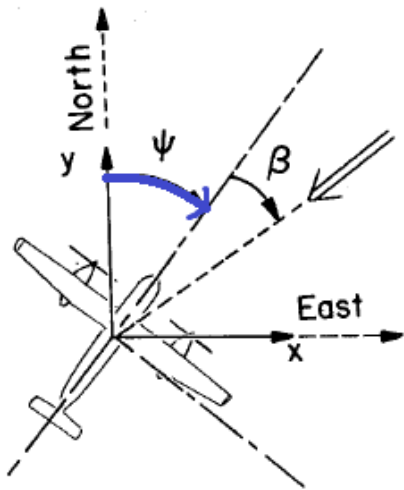
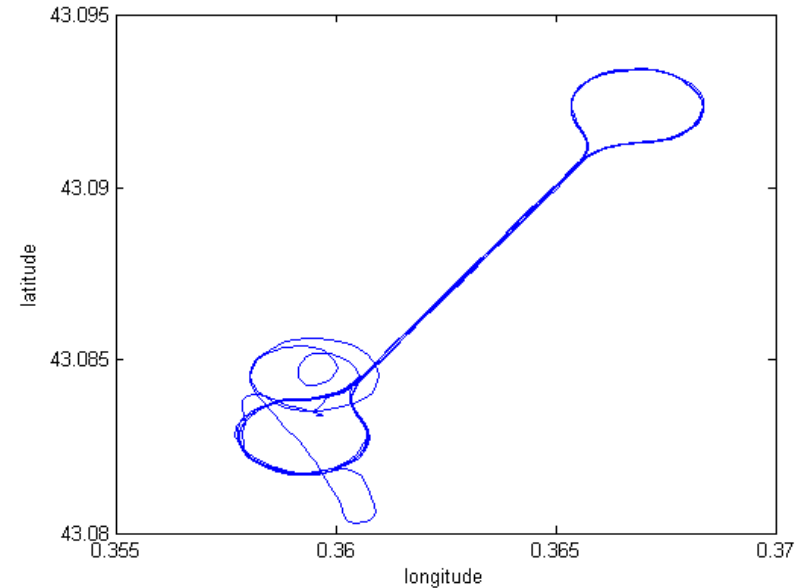
Tubing length variation

- Non-turbulent: similar behavior
- Turbulent: difference from around 10 Hz
- Similar for α and β



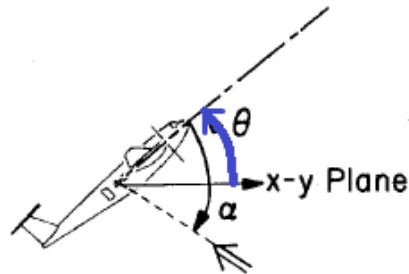
Theory and methods

- Synchronization of 5-hole probe and attitude information
- Select straight flight segments
- Calculate $\vec{v} = (u,v,w)$



TOP VIEW

← DIRECTION OF AIRSTREAM



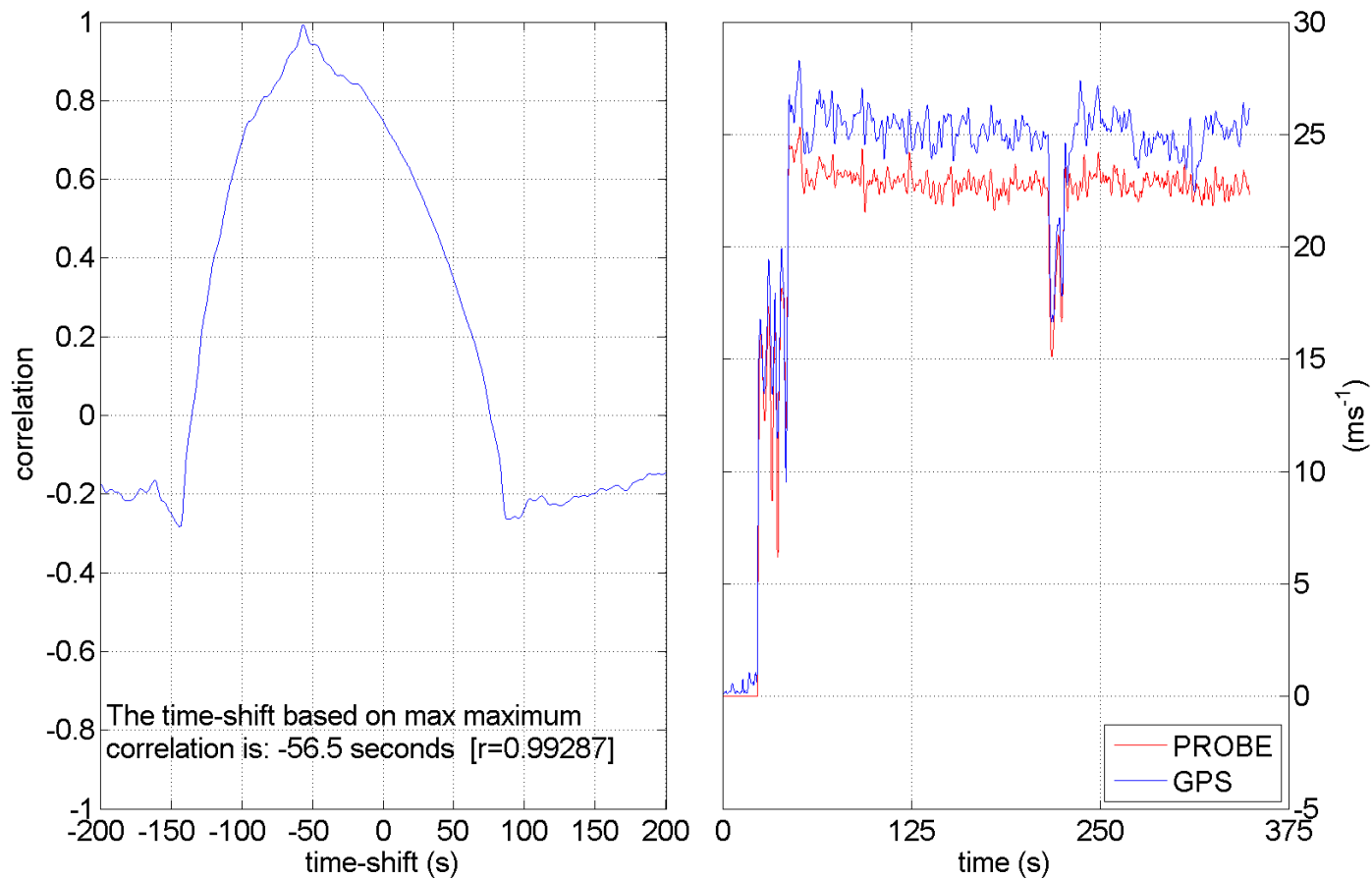
SIDE VIEW



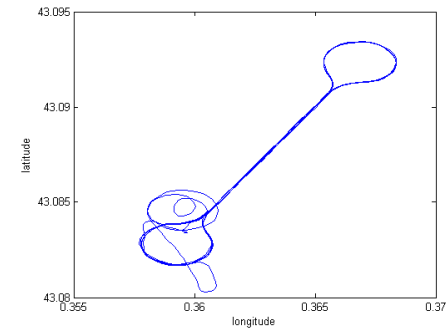
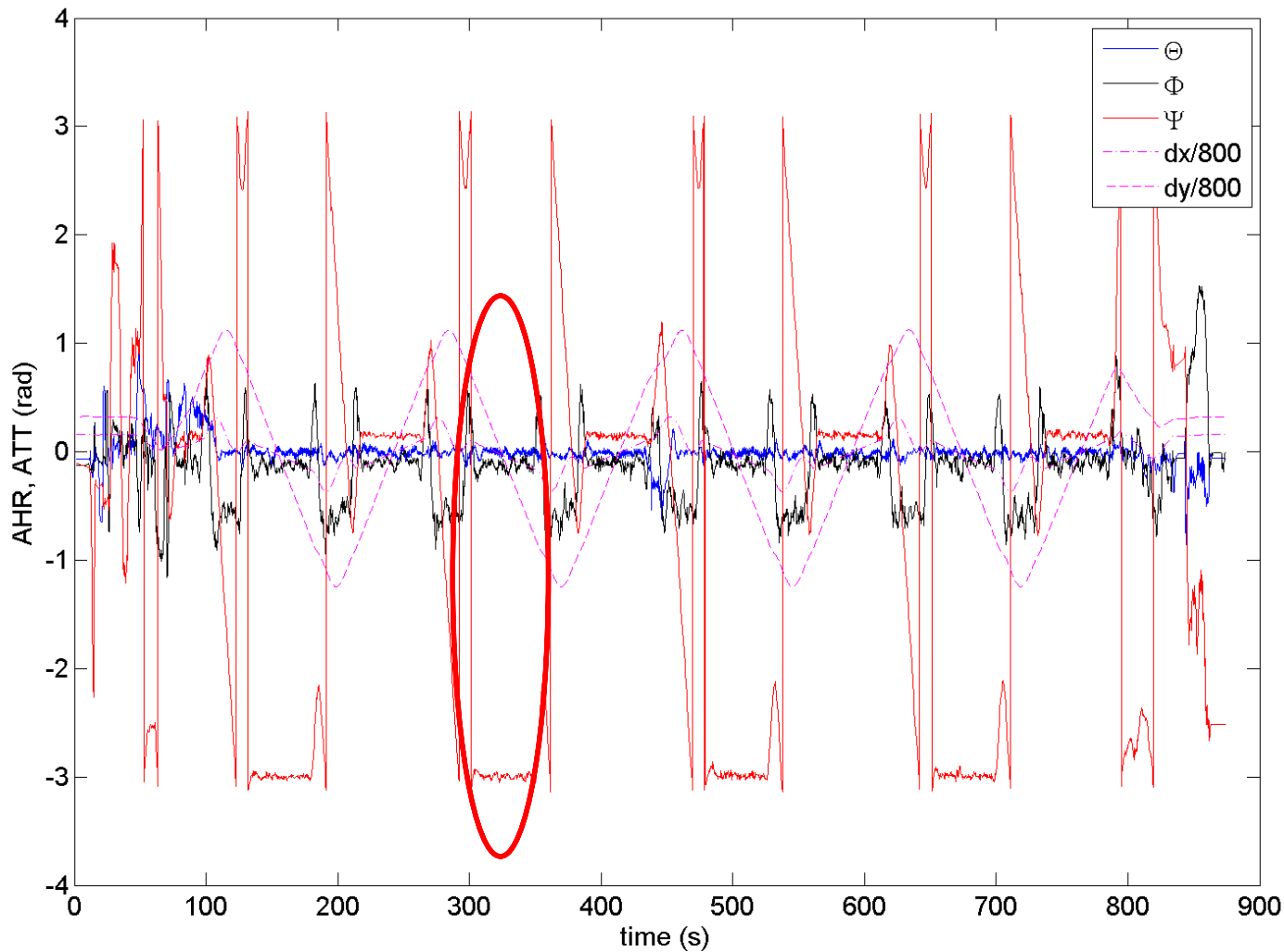
FRONT VIEW

Figure from Lenschow (1986)

Synchronization of 5-hole probe measurements and attitude information

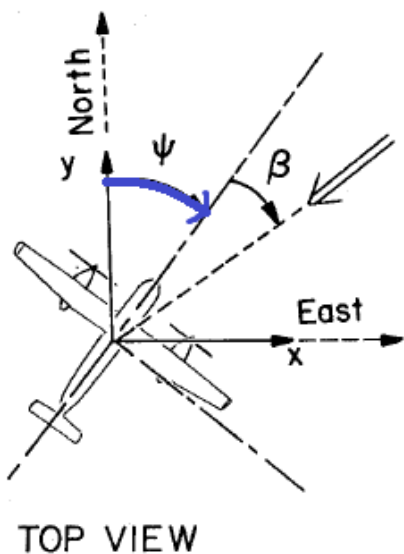
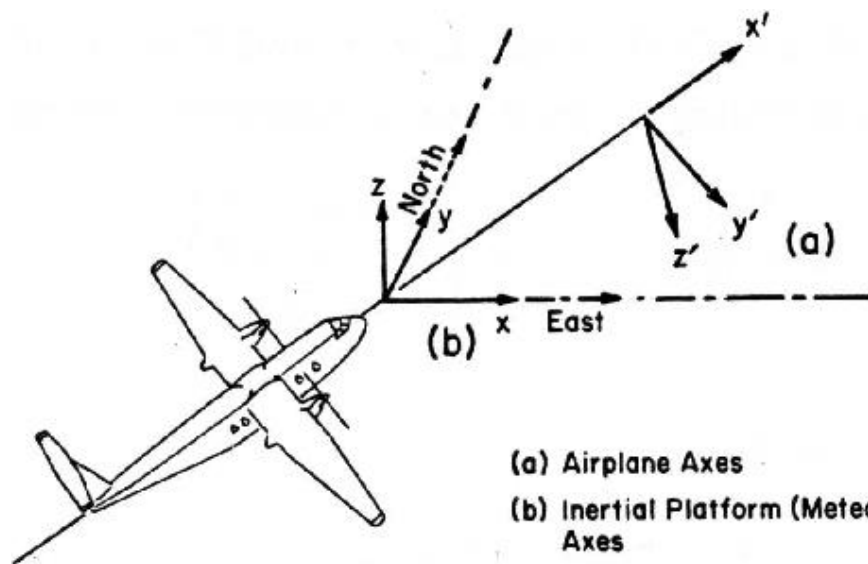


Selecting the straight flight segments

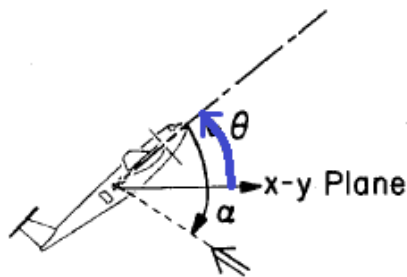


Calculating \vec{v}

- $\vec{v} = \vec{v}_p + \vec{v}_a$
- $\vec{v} = \vec{v}_p + \vec{v}_a + \vec{\Omega}_p \times \vec{R}$



← DIRECTION OF AIRSTREAM



Calculating \vec{v}

$$u = -UaD^{-1} \left(\begin{array}{l} \sin \Psi \cos \theta + \tan \beta (\cos \Psi \cos \Phi + \sin \Psi \sin \theta \sin \Phi) \\ + \tan \alpha (\sin \Psi \sin \theta \cos \Phi - \cos \Psi \sin \Phi) \end{array} \right) + u_p - L(\dot{\theta} \sin \theta \sin \Psi - \dot{\Psi} \cos \Psi \cos \theta)$$

$$v = -UaD^{-1} \left(\begin{array}{l} \cos \Psi \cos \theta + \tan \beta (\sin \Psi \cos \Phi - \cos \Psi \sin \theta \sin \Phi) \\ + \tan \alpha (\cos \Psi \sin \theta \cos \Phi + \sin \Psi \sin \Phi) \end{array} \right) + v_p - L(\dot{\Psi} \sin \Psi \cos \theta + \dot{\theta} \cos \Psi \sin \theta)$$

$$w = -UaD^{-1} (\sin \theta - \tan \beta \cos \theta \sin \Phi - \tan \alpha \cos \theta \cos \Phi) + w_p + L\dot{\theta} \cos \theta$$

$$u = -UaD^{-1} \sin(\Psi + \beta) + u_p$$

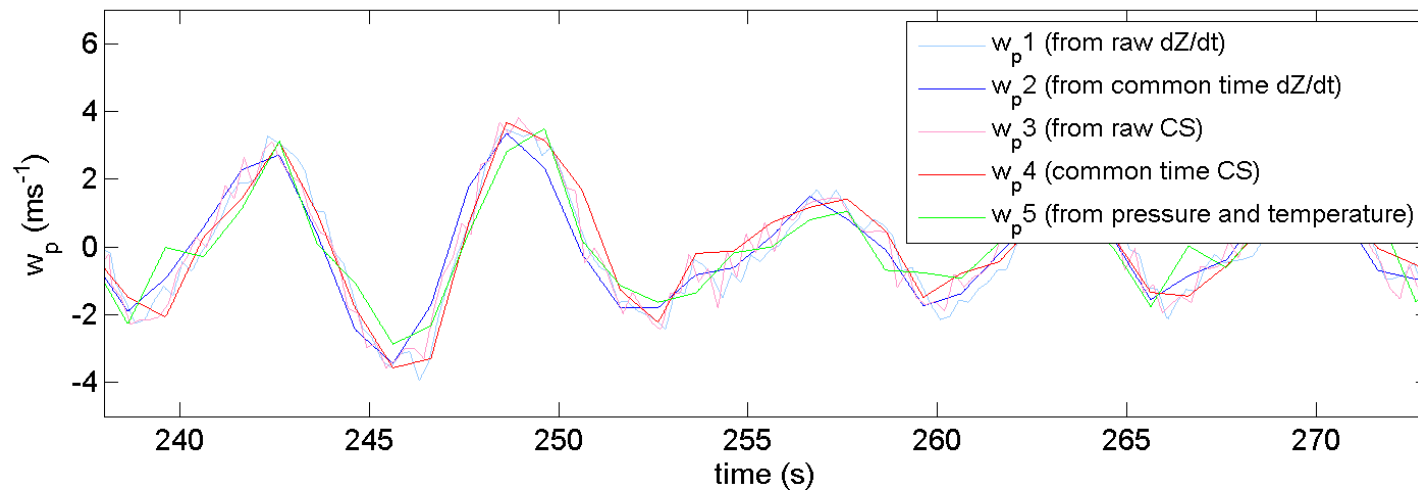
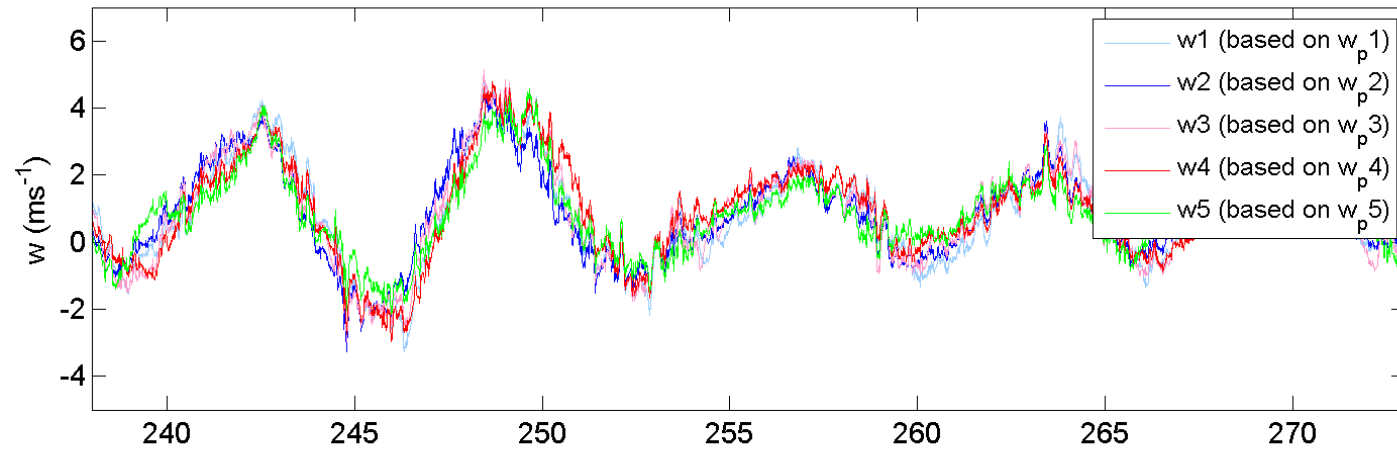
$$v = -UaD^{-1} \cos(\Psi + \beta) + v_p$$

$$w = -UaD^{-1} \sin(\theta - \alpha) + w_p$$

$$D = \sqrt{1 + \tan^2 \alpha + \tan^2 \beta}$$

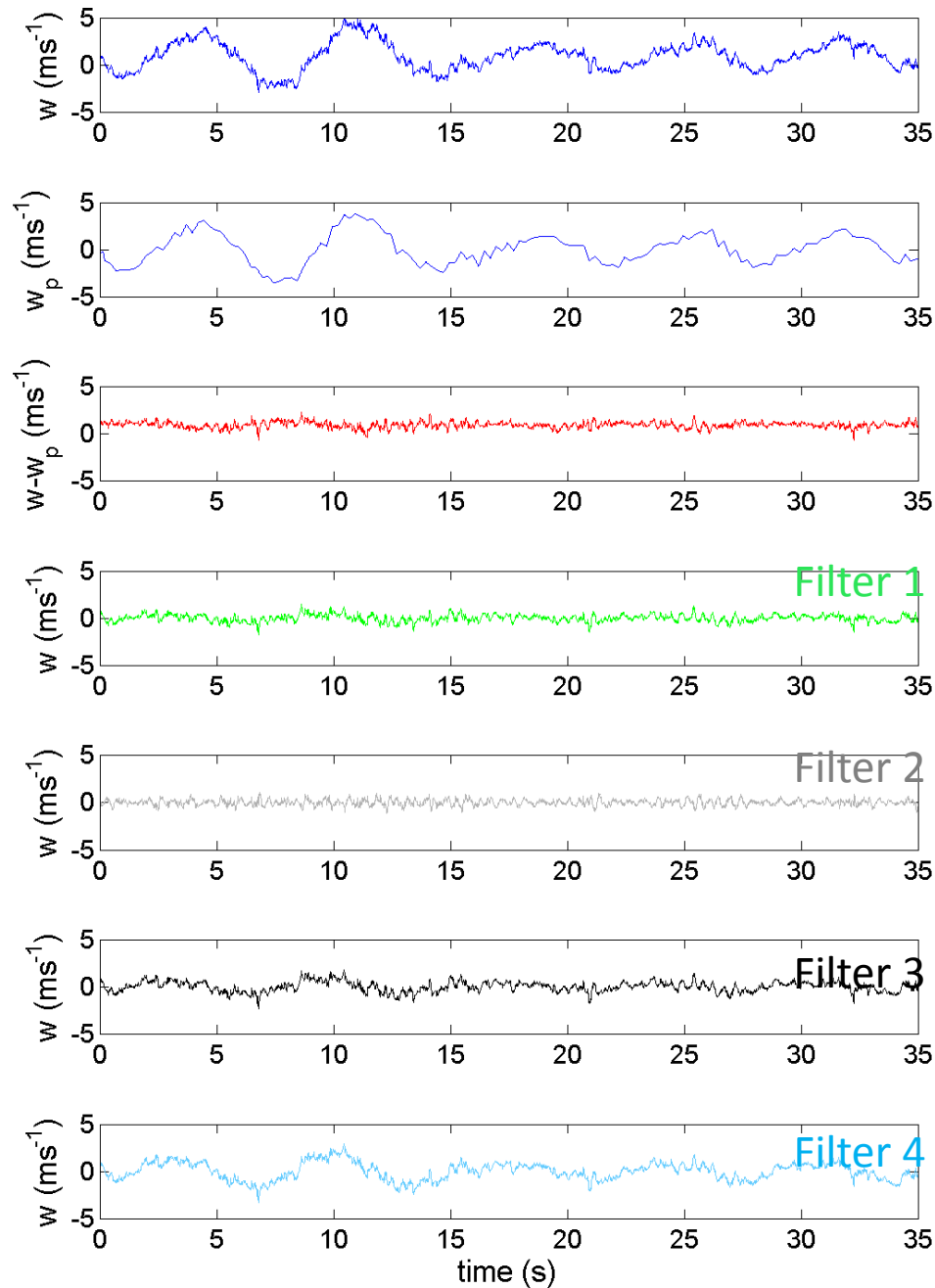
Calculating \vec{v}

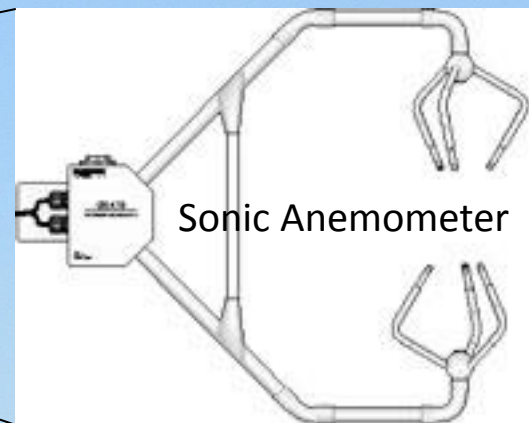
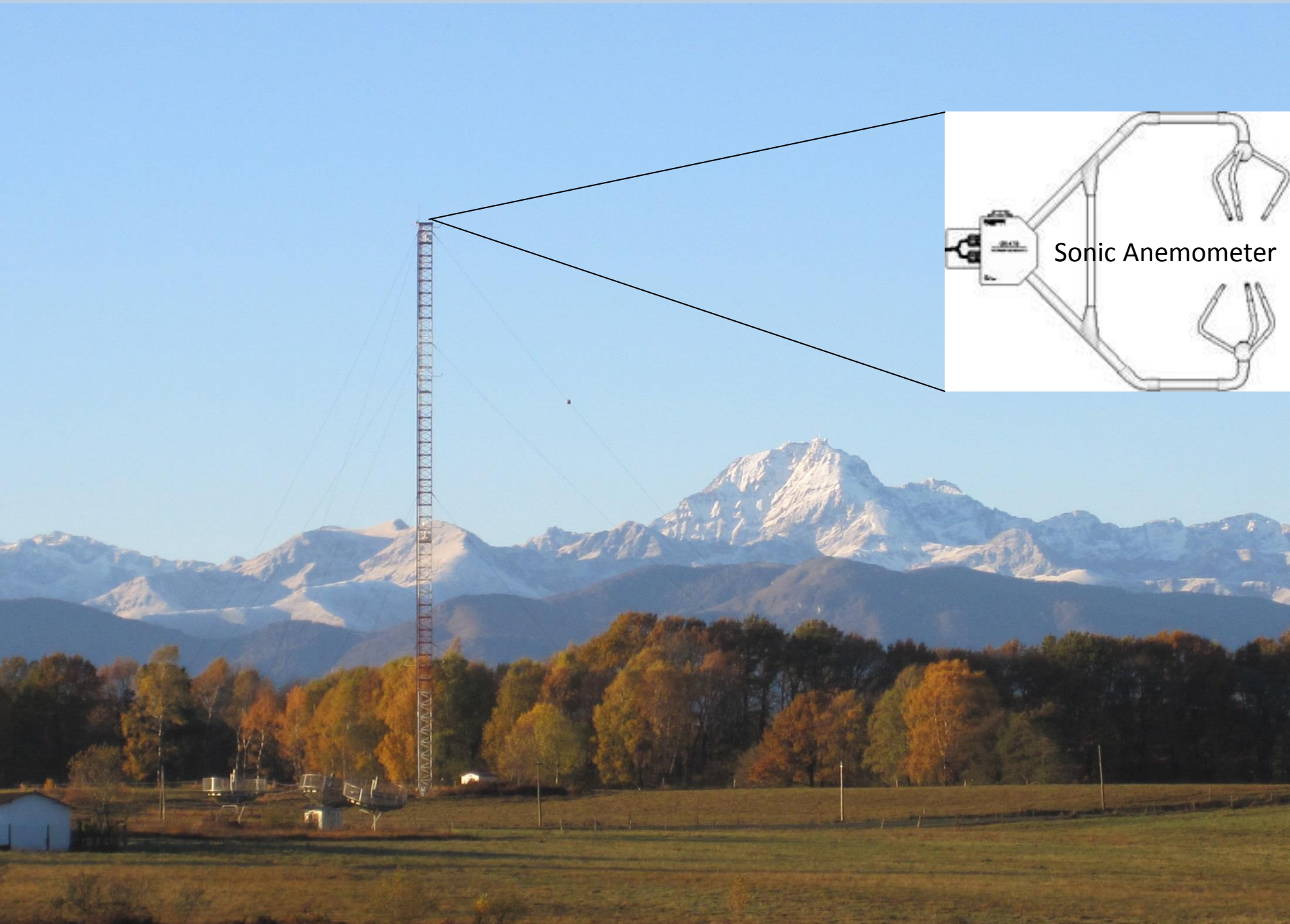
$$w = -UaD^{-1} \sin(\theta - \alpha) + w_p$$



Calculating \vec{v} : Highpass filter

- Remove \vec{v}_p
- Butterworth filter
- Different cutoff frequency and filter order



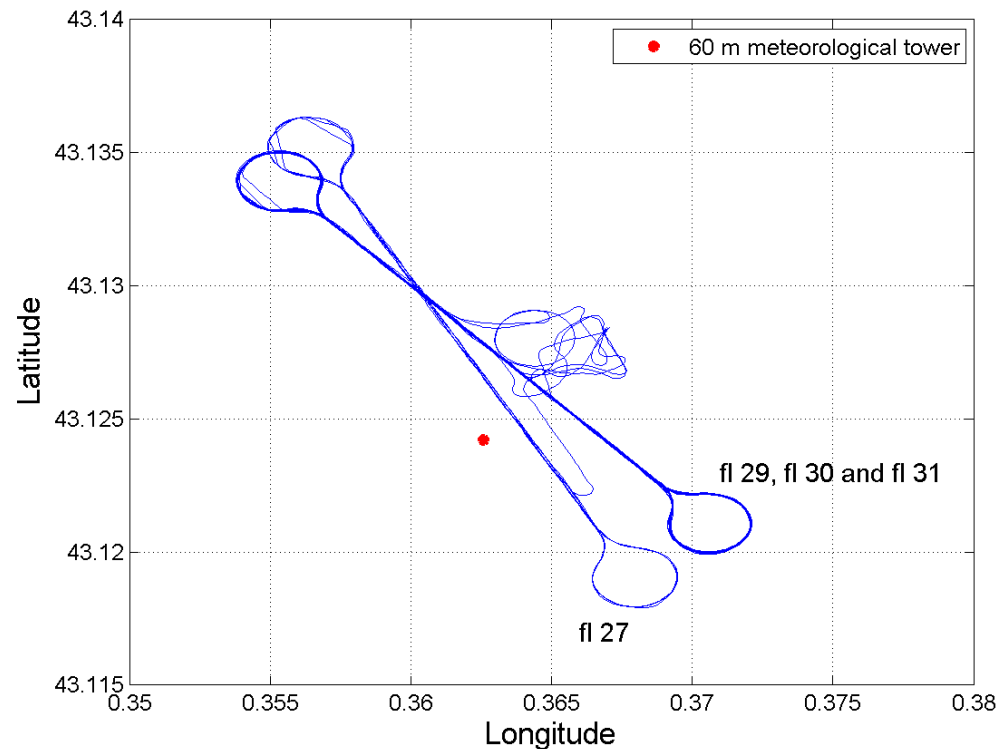


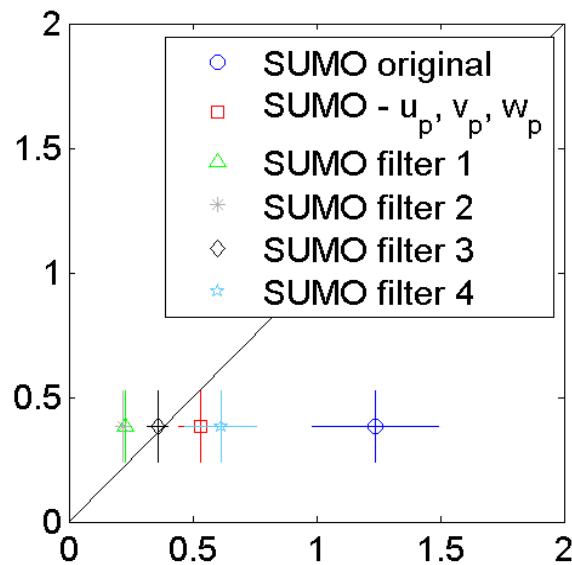
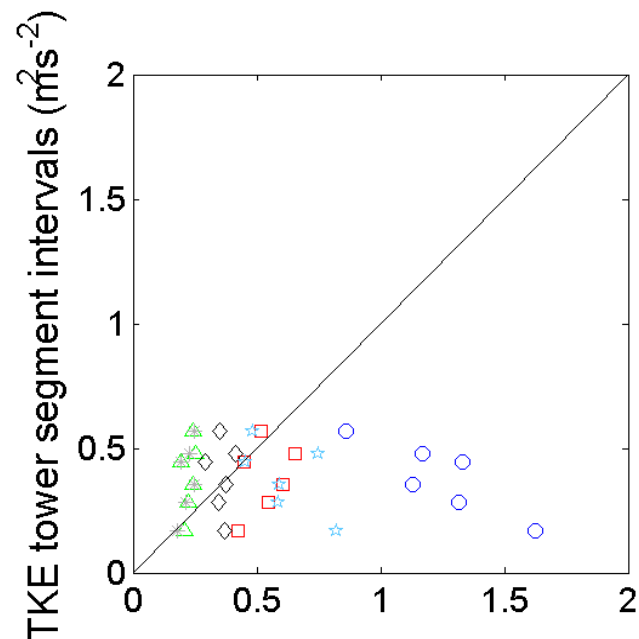
Comparison to a 60 m meteorological tower

- 17 flights in the vicinity of the tower
- 4 flights available for comparison (26 segments)

- $$\frac{TKE}{m} = \frac{1}{2} (\overline{u'^2} + \overline{v'^2} + \overline{w'^2})$$

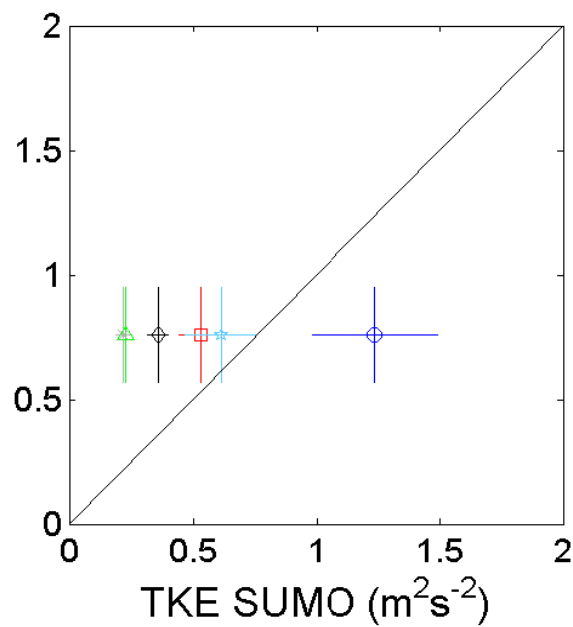
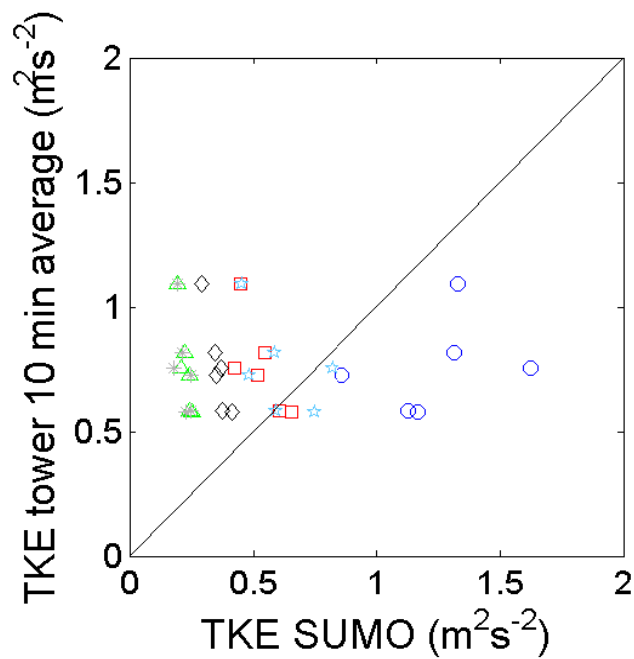
- $$u^* = (\overline{u'w'^2} + \overline{v'w'^2})^{1/4}$$

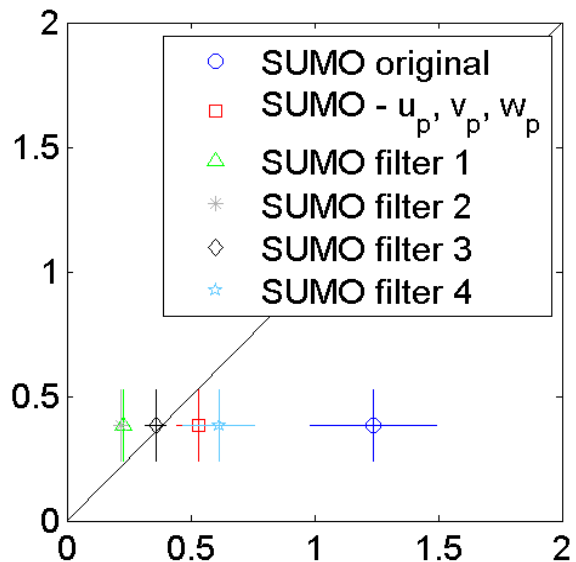
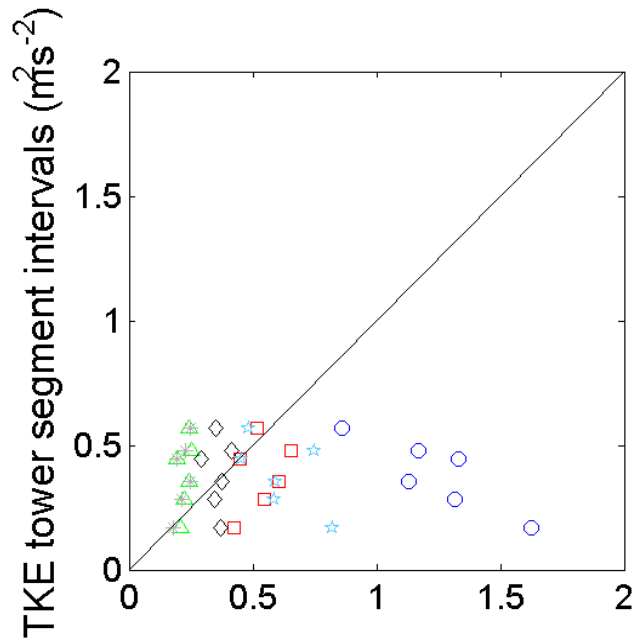




TKE
FL 29

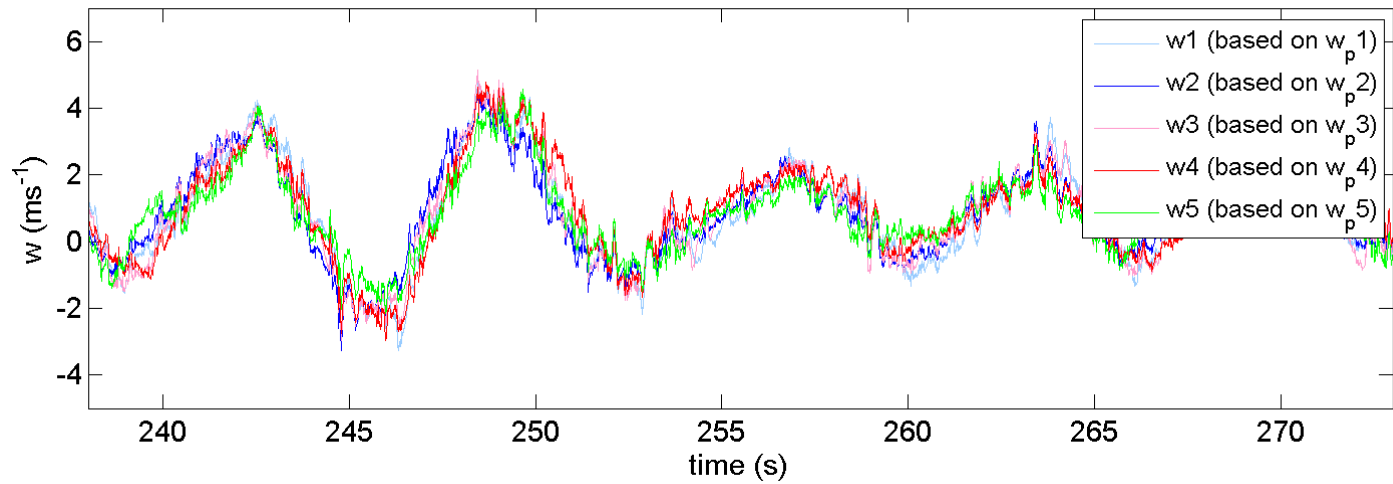
- 6 straight flight segments

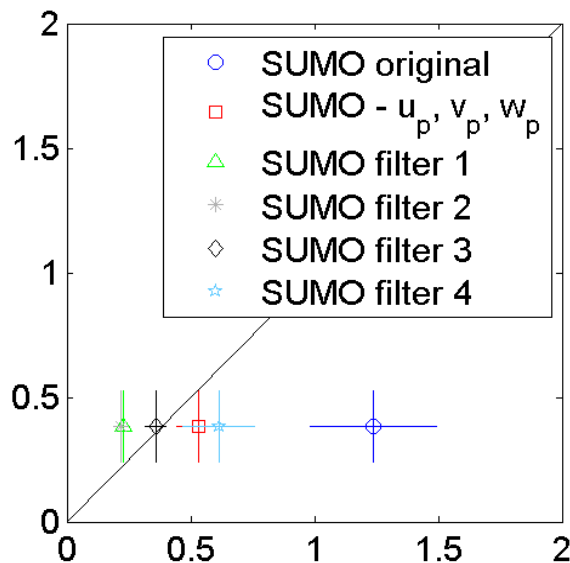
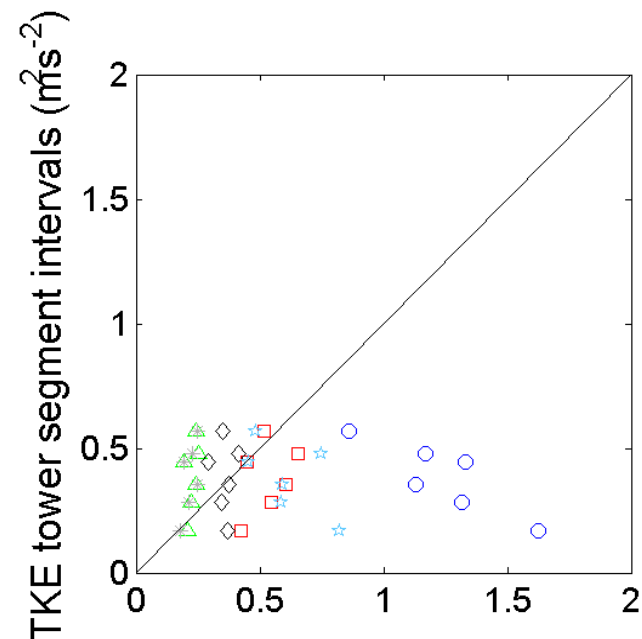




TKE
FL 29

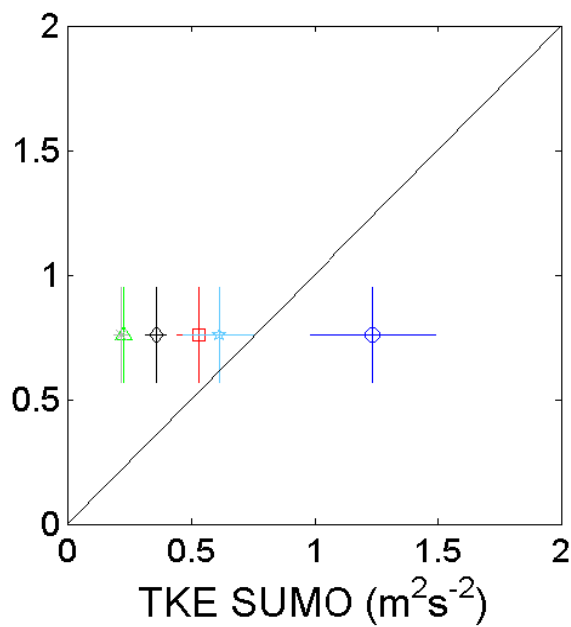
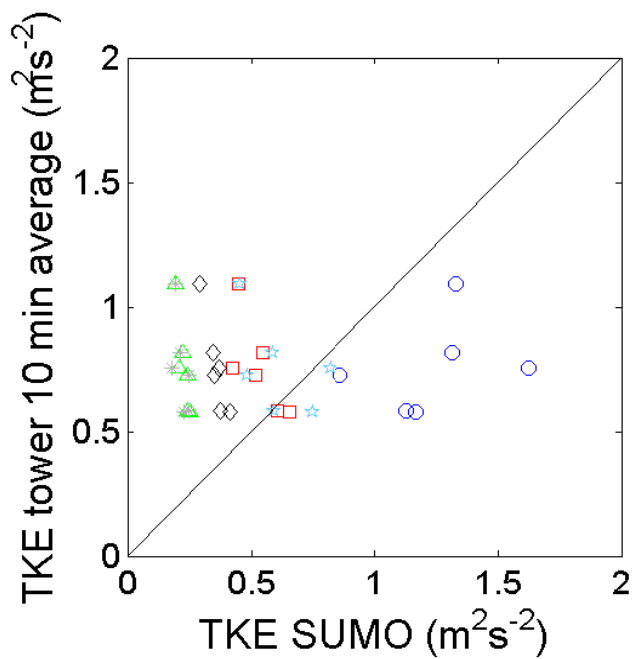
- 6 straight flight segments

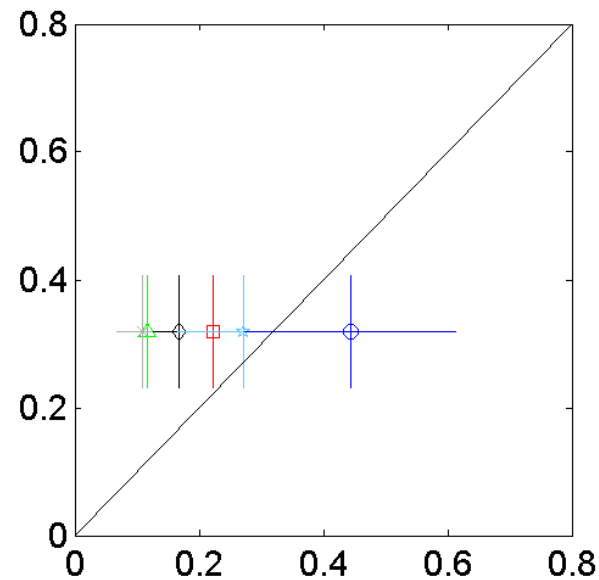
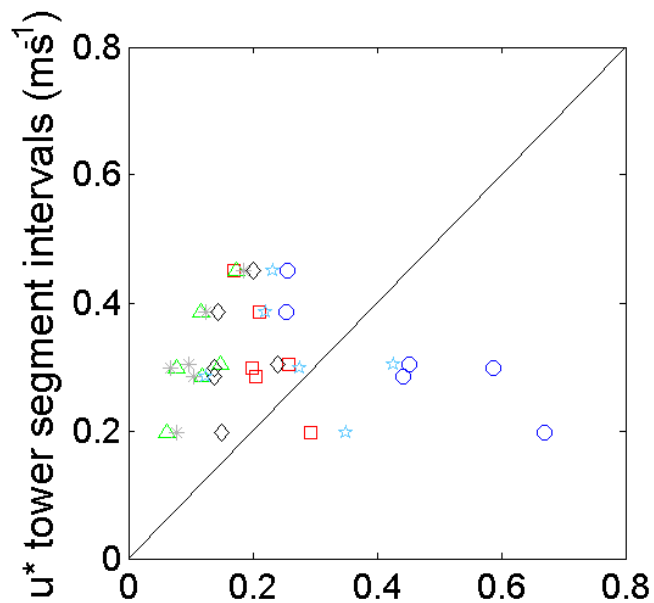




TKE
FL 29

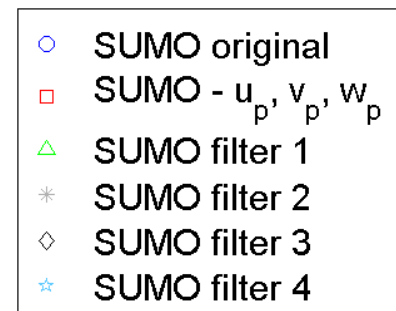
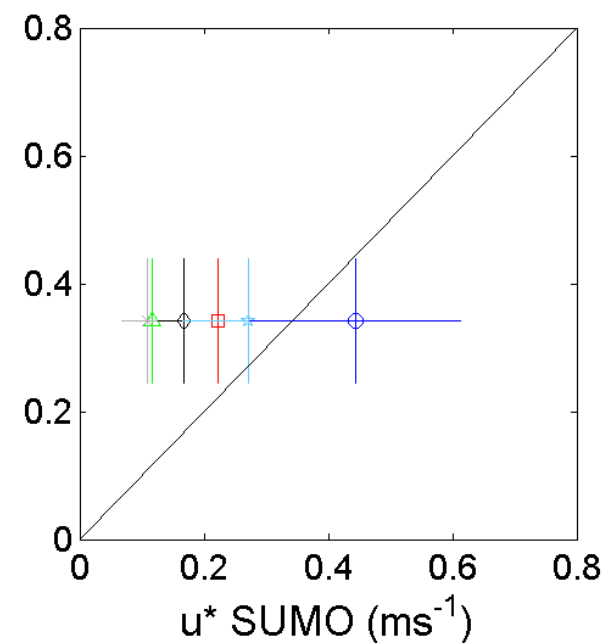
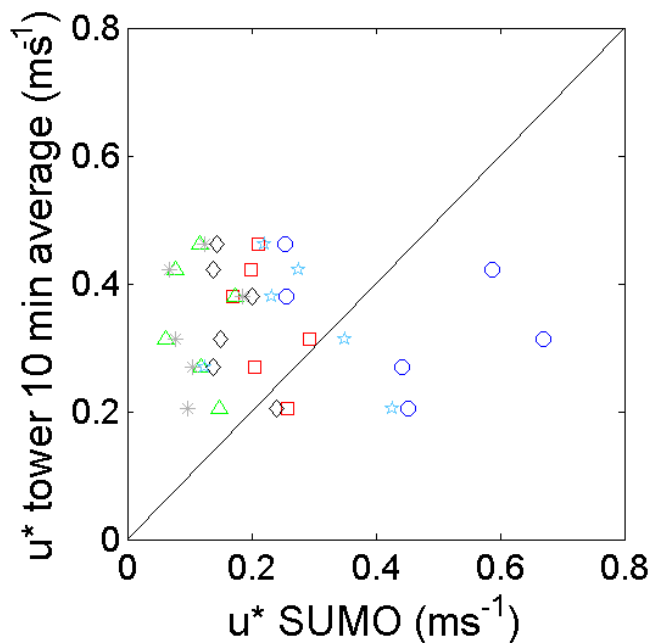
- 6 straight flight segments
- Filter 3





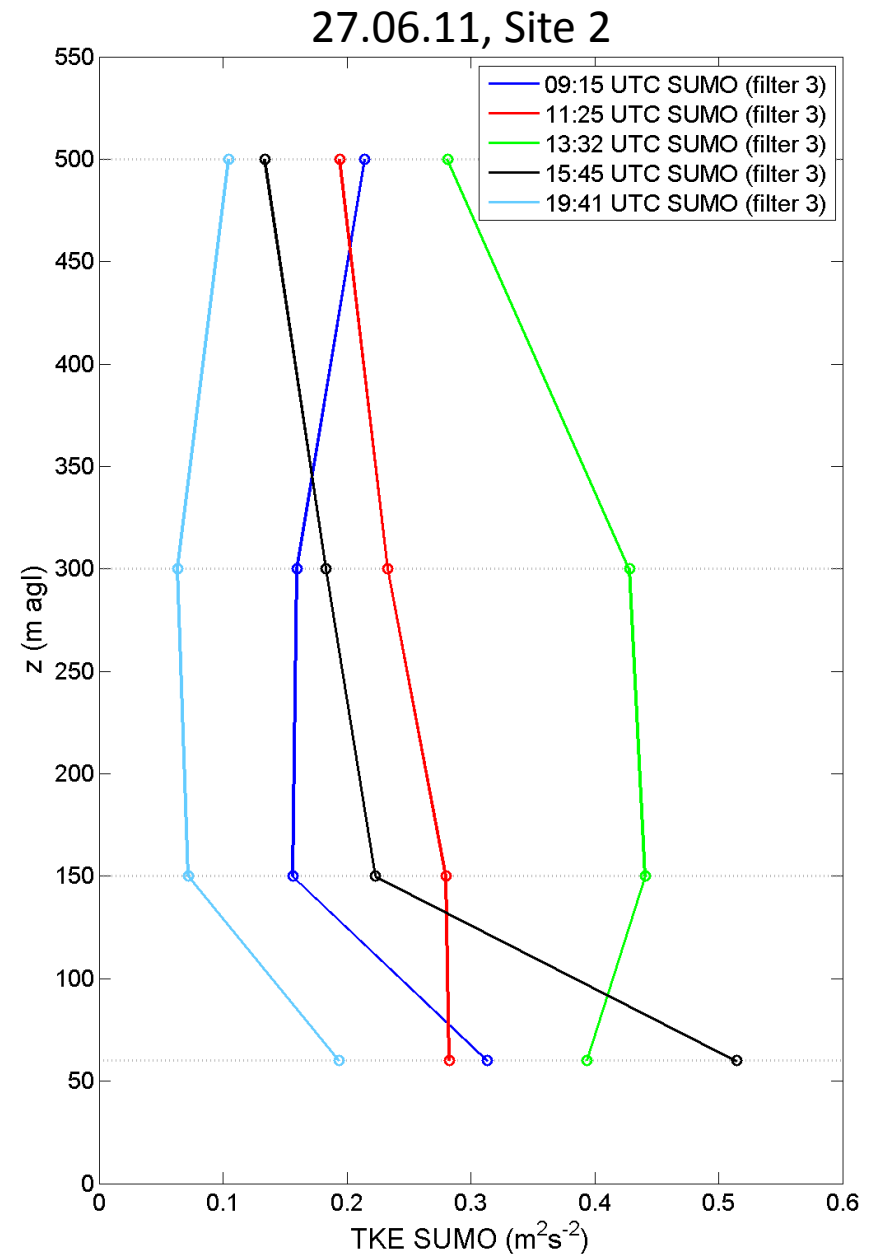
u^*
FL 29

- 6 straight flight segments
- Filter 4



TKE profiles

- Low TKE in morning/evening
- Maximum early afternoon
- Daily evolution mainly driven by buoyancy
- Shear production close to the ground

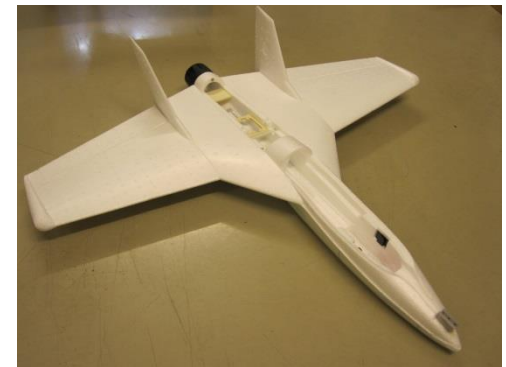


Summary

- A new 5-hole probe sensor
- Wind tunnel testing
 - 5-hole probe able to react to turbulence (~ 20 Hz)
 - Tubing of 15 cm
- BLLAST
 - SUMO overestimate TKE and u^* due to aircraft movement
 - Highpass filter improve the results
 - SUMO capture daily TKE evolution by profiles

Future work

- New version: 5-hole probe and attitude on common datalogger with 100 Hz temporal resolution
- Wind tunnels tests with higher turbulence intensity (whole airframe due to flow distortion)
- In-flight calibration maneuvers to detect offsets in 5-hole probe measurements of angles/airspeed
- Small test campaign under planning: SUMO mounted on a car together with 2 sonic anemometers
- Campaign in the vicinity of wind turbines together with a Lidar (Netherlands)

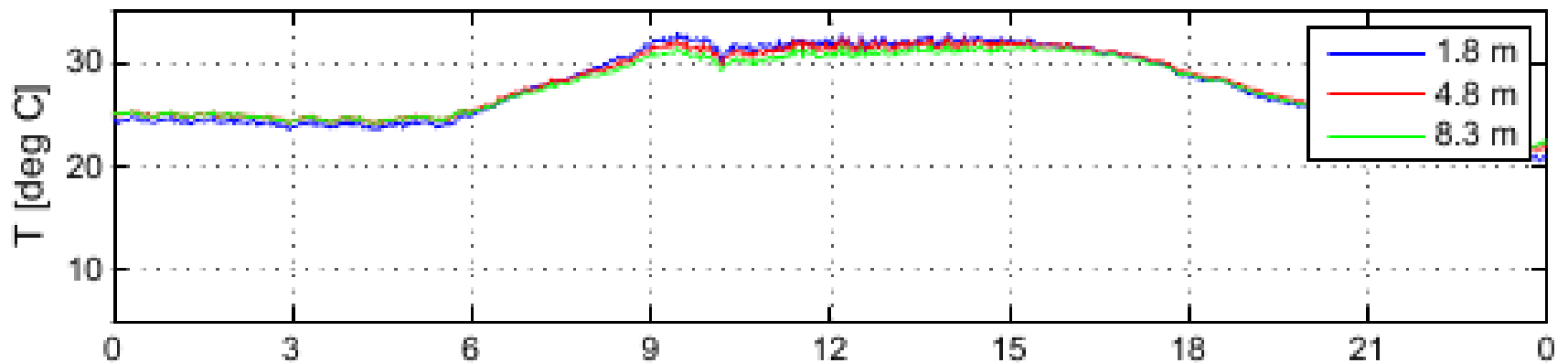
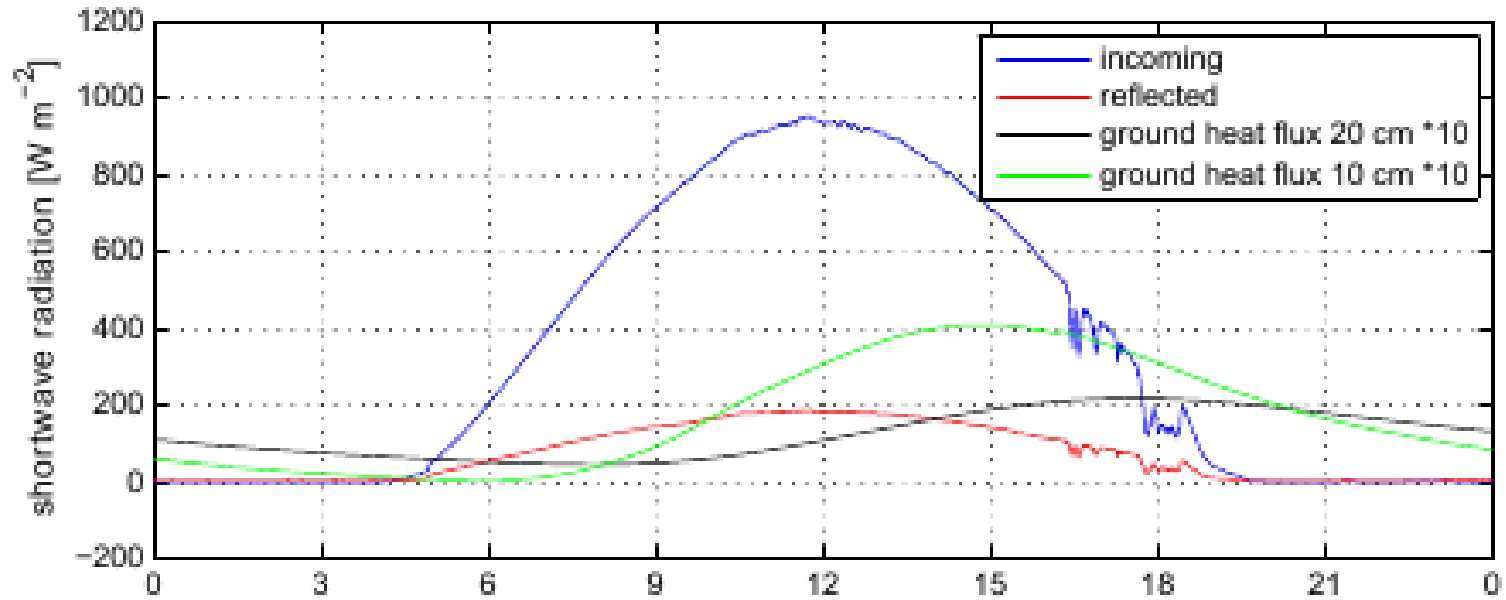


Thank you!

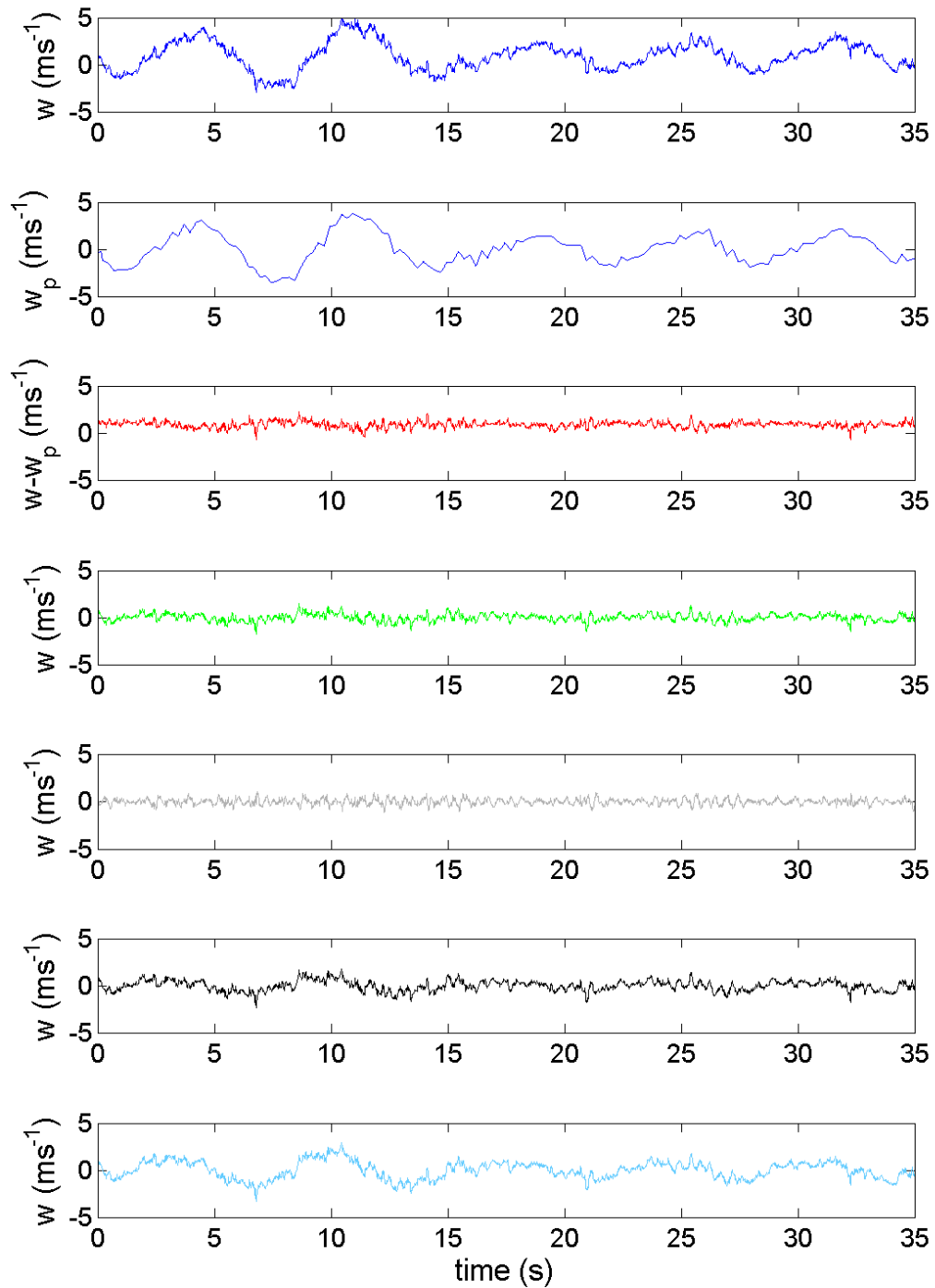


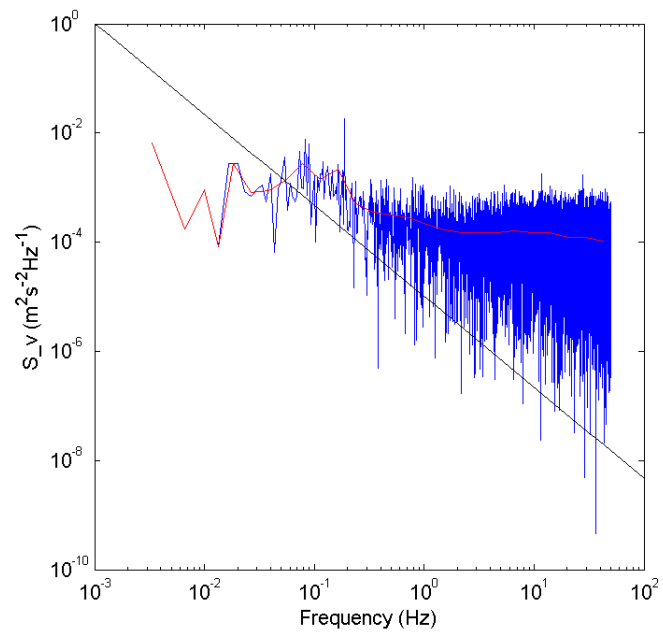
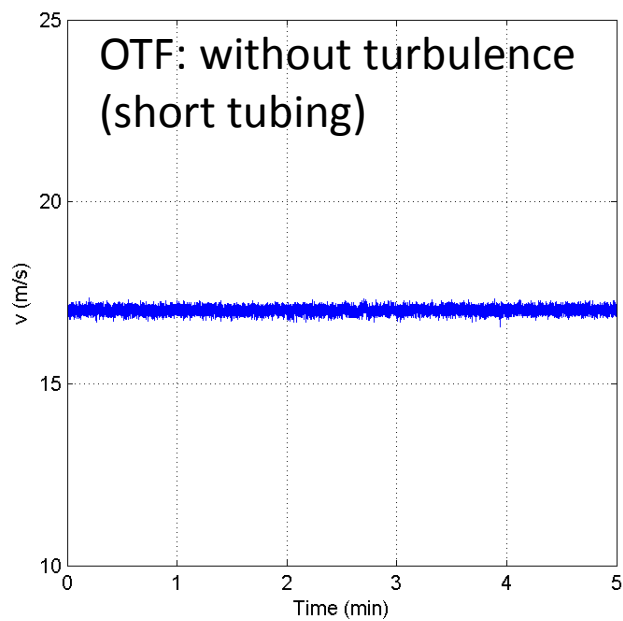
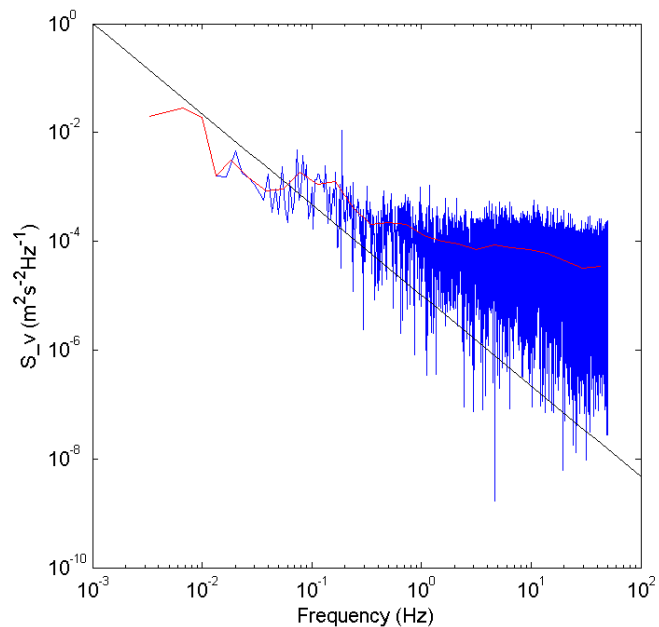
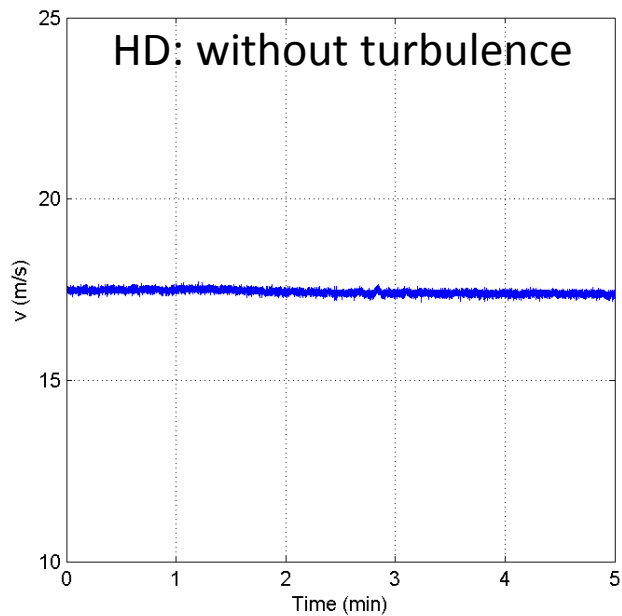
Picture by Jesper Hjelme, IWAL

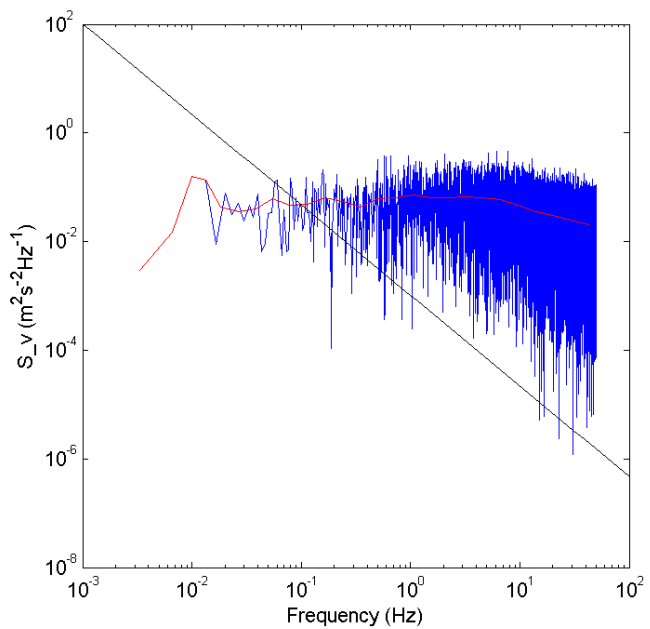
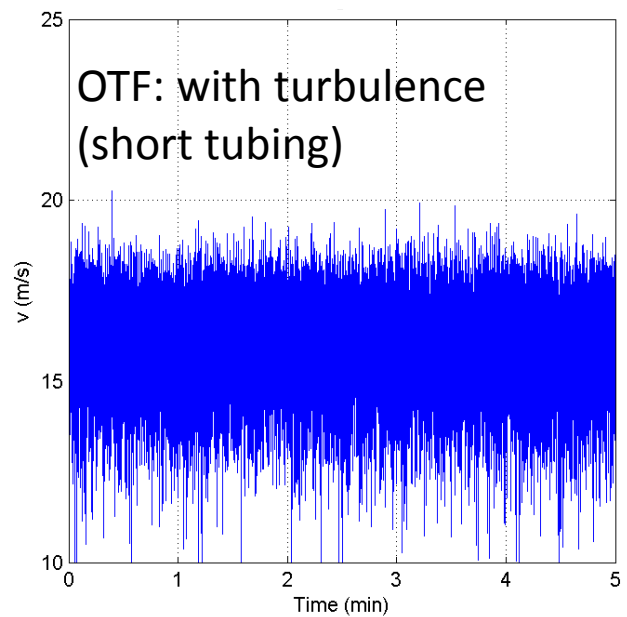
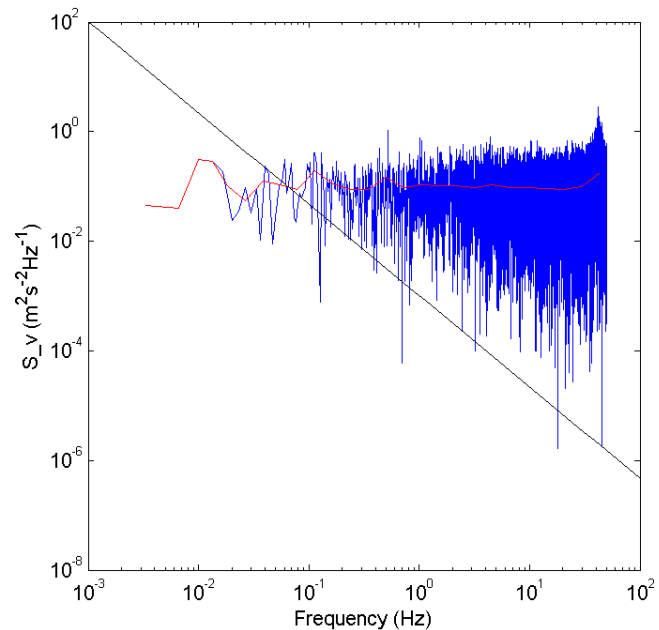
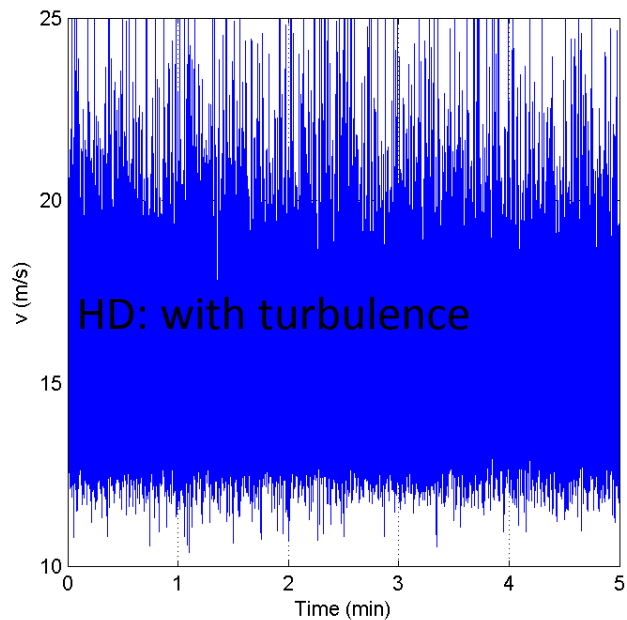
Meteorological conditions at Site 1, 27.06.11



- Filter 1 (N=1, $W_n=1$ Hz)
- Filter 2 (N=10, $W_n=1$ Hz)
- Filter 3 (N=1, $W_n=0.5$ Hz)
- Filter 4 (N=1, $W_n=0.2$ Hz)







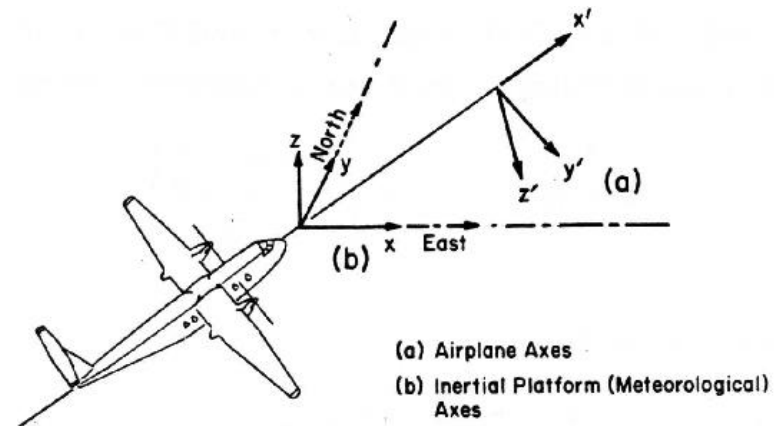
Calculating the velocity of the aircraft with respect to the ground (\vec{v}_p)

1 (dx/dt, dy/dt and dz/dt)

X = GPS_utmE

Y = GPS_utmN

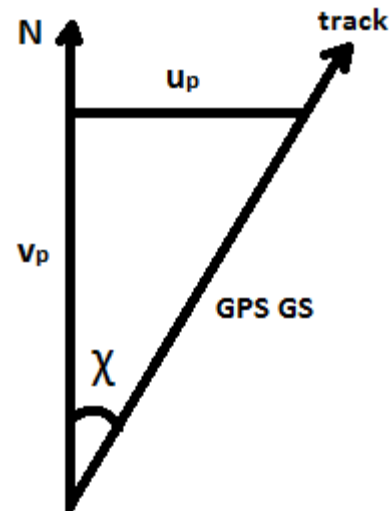
Z = altitude



2 (GS, CS, χ)

3 (temperature SHT_T and pressure SCP_P)

$$W_p = \frac{dz}{dt} = - \frac{RT}{pg} \frac{dp}{dt}$$



$$Up = GPS_GS * \sin \chi$$

$$Vp = GPS_GS * \cos \chi$$

$$Wp = GPS_CS$$

